



	Material Technology Cathodes, Gimbals	Field Emission / Colloid Thrusters	Hall Thrusters 1	Hall Thrusters 2
	HS6	SR5	SR6	SR2
15.15	A865 Comparisons between Particle Simulation using IAT Growth Model and Plume Measurements of LaB6 Hollow Cathode <i>K. Kubota</i>	A317 Uncertainty Quantification of Electro spray Thruster Array Lifetime <i>B. Jorns</i>	A660 I2HET: Development of an Iodine-Fed Hall Effect Thruster <i>M. M. Saravia</i>	A648 Multiscale Hybrid Modeling of Unconventional Hall Thrusters <i>M. Laterza</i>
15.30	A874 Fluid model of a Hollow Cathode discharge <i>X. Chen</i>	A526 Colloid Thruster Plume Simulations Using a Particle-Particle Model <i>J. Wang</i>	A838 Simulation of an Adaptable Hall Thruster <i>N. Proulx</i>	A281 Plasma Simulations for the Assessment of Pole Erosion in the Magnetically Shielded Miniature Hall Thruster (MaSMi) <i>A. Lopez Ortega</i>
15.45	A153 Development of Low-Voltage-Driven Propellantless Cathodes with High-Current Density Based on Graphene-Oxide-Semiconductor Structure <i>R. Furuya</i>	A530 Effect of Aprotic + Protic Mixtures on Electro spray Droplet Fragmentation <i>D. Levin</i>	A902 Performance, Plume, Stability, and Wear Characterization of Three Alternate Magnetic Field Topologies in the Hall Effect Rocket with Magnetic Shielding <i>H. Kamhawi</i>	A364 Particle-In-cell / fluid simulations of a Hall effect thruster accounting for plasma wall interactions <i>W. Villafana</i>
16.00	A214 Radio Frequency Microdischarge Neutralizer <i>F. Filleul</i>	A830 Development of an Electro spray Source Model for Kinetic Plume Modeling <i>E. Petro</i>	A637 Characterisation and Performance Comparison of a Low-Power Hall-Effect Thruster and an Advanced Cusp Field Thruster with Multiple Noble Gases <i>M. Vaupel</i>	A478 The Sputtering Mechanism of Keeper Electrode in Hall Thruster <i>J. Chen</i>
16.15	×	A565 Numerical simulation of electro spray thruster extraction for highly conductive propellants <i>H. Huh</i>	A752 Preliminary Evaluation of Anode-Layer-Type Hall Thruster Performance Using Pulsating Boost Chopper Circuit <i>K. Nagamine</i>	×
16.30	×	A559 Study on Performance of Ionic Liquid Electro spray Thruster in Atmosphere and Vacuum Environment <i>Y. Guo</i>	A563 Impact on Performance and Erosion in Hall Thruster using Argon and Xenon propellant <i>S. Yokota</i> <i>K. Shimamura</i>	×
16.45	×	A788 High-Speed Transient Characterization of the Busek BET-300-P Electro spray Thruster <i>D. Courtney</i>	×	×
17.00	×	×	×	×

Session End → Gala Dinner beginning 18:30

Ion Thrusters	Resistojets/ Arcjets	Innovative Concepts	Thruster Concepts
HS5	SR4	HS3	HS2
A680 Surface Modification of Pyrolytic-Graphite Grids for an Ion thruster <i>Y. Matsunaga</i>	A837 Numerical Rebuilding of Very Low Power Arcjet Thruster VELARC in Test Facilities at IRS and ESA-ESTEC <i>P. P. Upadhyay</i>	A646 Operation and Performance of a Fully-Integrated ion Electro spray Propulsion System <i>B. Kristinsson</i>	A537 An Experimental Study of Thrust Dependence on Magnetic Field in an Electrodeless Inductive Plasma Accelerator <i>A. Tatsuno</i>
A722 Uncertainty Quantification of Modeled Electron Backstreaming Failure for the NEXT Ion Thruster <i>J. Yim</i>	A562 A Chemically Augmented Arcjet Thruster with Exotic Propellants <i>M. Tsuchiya</i>	A742 Direct Inertial Electrostatic Confinement Propulsion at Low Power Levels <i>M. Winter</i>	A596 Development and Characterization of the Helicon Plasma Thruster Prototype HPT-03 <i>J. Navarro-Cavallé</i>
A907 Modeling Ion Optics Erosion in the NEXT Ion Thruster Using the CEX2D and CEX3D Codes <i>J. Polk</i>	A703 Effect of adding water to propellant of a DME arcjet thruster <i>T. Tachibana</i>	A761 Experimental demonstration of thrust vectoring magnetic nozzle with multi-axis thrust measurement system <i>M. Edamoto</i>	A647 Optimization of electrothermal microwave plasma thruster for nanosatellites <i>S. Ivanov</i>
×	×	×	A855 Performance improvement of a magnetic nozzle helicon plasma thruster <i>K. Takahashi</i>
×	×	×	A867 Cubesat Test Platform for miniaturized electric propulsion system verification campaign <i>F. Stesina</i>
×	×	×	A815 Enabling High-Energy Missions with Nanosatellites by Using Ablative Pulsed Plasma Thrusters <i>P. Gessini</i>
×	×	×	A920 The operation of a low-power cylindrical Hall thruster with zinc as the propellant <i>C. Ryan</i>
×	×	×	A746 On the Performance of Arcjet Thrusters using Numerical Modeling: A case study of Hydrogen as a propellant <i>D. Akhare</i>

